



Flight Software

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Flight Software Topics



Flight Software Functional Requirements

Flight Software Interfaces

Flight Software Architecture

Software Development

Software Testing

Software Configuration Management

Flight Software Builds and Schedule



Bus Flight Software Re- scope Impacts



- No Functional Changes Identified
- No Requirements Modified that Significantly Impact FSW Performance



Approach Re-Scope Impacts



- Minimize Cost
 - Maximize Flight Software Code Reuse (Primarily ICM)
 - Eliminate non-flight (one-off) code whenever possible
 - Reduce documentation effort as much as possible, especially during design fluctuations
- Maintain Schedule
 - Prioritize Design and Development of Flight Software based on least likely impact from other ongoing design trades

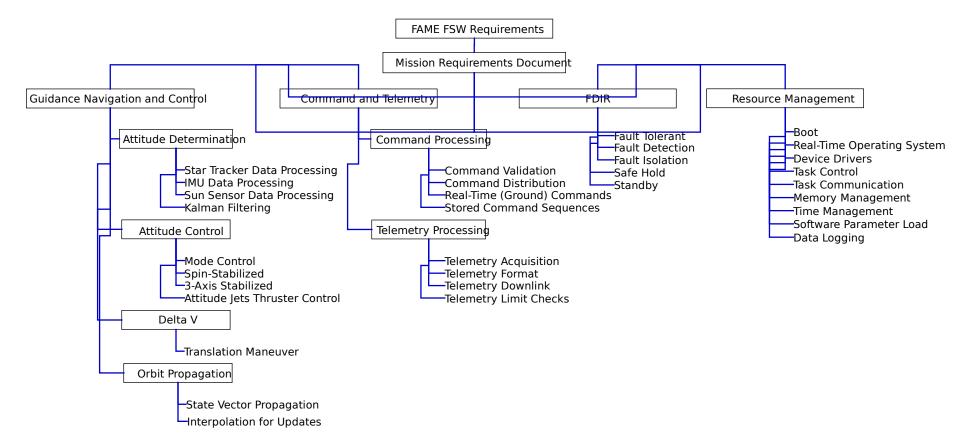


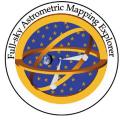
Top Level Requirements











Attitude Determination Requirements



- Acquire initial attitude with Star Tracker (ST) quaternion output processing
- Process Inertial Measurement Unit (IMU) data to update attitude estimates between Star Tracker updates
- Process Spinning Sun Sensor (SSS) and IMU data during Apogee Kick Motor (AKM) Maneuver for Auto Nutation Control (ANC)
- Provide selectable Kalman Filtering of ST, IMU, and SSS for spacecraft bus attitude estimates
- Provide time tagged best estimate attitude to the instrument at 1 second intervals (including attitude quaternion, body rates, earth and sun vectors)
- Provide time tagged best estimate attitude to the ground, including current mode and validity at a commandable rate
- Include instrument attitude information for further refinement when available
- Provide sanity checks on all sensors and attitude estimates
- Allow for ground intervention in setting, resetting, and selecting attitude sensors and attitude information



Attitude Control Requirements



- Perform seven attitude control modes
 - Standby, Inertial Pointing, Rate Control, Safe Hold, Open Loop Burn, Active Nutation Control, and Spin Axis Precession
- Monitor attitude and attitude rates for performance within defined error thresholds and report status
- Provide sanity checks on all commanded and autonomous attitude control actions
- Allow for ground intervention in setting, resetting, and limiting attitude control modes and attitude control information
- Log all commanded and autonomous attitude control mode changes



Maneuver Functional Requirements



- Provide Spin Axis Precession and Auto Nutation Control during AKM spinup, firing, and spindown maneuver
- Monitor Attitude Errors and Abort Maneuver if attitude errors exceed defined thresholds
- Minimize probability of system reset during this mission critical phase
- Minimize system reset/restart time in the unlikely event of failure to retain or regain spacecraft stability within the time to effect
- Validate and Process Uplinked Maneuver Parameters
- React to real-time commands to cancel maneuver
- Log all executed maneuver commands and telemetry



Navigation Functional Requirements



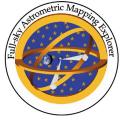
- Allow for active ranging with the ground station to be performed at commanded intervals (which denies the use of the transponder for FSW activities) without the loss of stored science data
- Process uplinked state vector tables and perform on-board orbit propagation calculation
- Provide updated state vector information as input to the Attitude Determination process
- Provide Sun and Earth vectors to the instrument as required
- Log orbit determination products
- Provide orbit determination products as telemetry to the ground



Command Functional Requirements



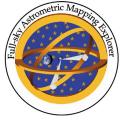
- Support CCSDS Command Data Format
- Authenticate Received Commands (Time, Authentication Count)
- Validate Command Data Packets (Checksum)
- Distribute Real-Time Commands to Application via the Software Bus
- Store Time-Tagged (Absolute and Relative) and Event-Driven Commands
- Issue Time-Tagged Commands Based on System Time
- Issue Event-Driven Commands Upon Receipt of Significant Event
- Provide the Capability to Enable/Disable Command Sequences
- Log Real-Time, Time-Tagged, and Event-Driven Commands as Executed
- Provide Confirmation of Command Execution in Telemetry



Telemetry Functional Requirements



- Support CCSDS Telemetry Data Format
- Acquire Spacecraft Bus Telemetry
- Store and Limit Check Selected TM Items Based on TM Database/Current Value Table (TDB/CVT)
- Monitor TDB/CVT and Generate an Event/Alarm for TM Items Exceeding Specified Limits and Persistence Counters
- Log All Events and Alarms
- Provide Sticky Indicators for Identification of Triggered Events and Errors
- Format Real-Time Telemetry for Downlink
- Time Tag Science Data for Downlink
- Rate Limit Telemetry Downlink Based on Selected Downlink Mode
- Log Selected CCSDS Telemetry Packets to Telemetry Log
- Load and Dump the TDB/CVT Parameters
- Dump Log Packets (Based on Configurable Parameters)
- Provide Extended Diagnostics Mode for Increased Rate Telemetry



Resource Manager Requirements



- Provide a Real-Time Operating System and Application Program Interface
- Program Boot PROMs for Diagnostics, multiple EEPROM Software Load options, and a Minimal Command and Telemetry Capability
- Provide a Software Bus for Routing CCSDS Packets within FAME Applications
- Provide 1553 Bus Controller for RIUs and Instrument C&T Interfaces
- Provide Memory (Software/Data) and Object (Structured Table)
 Load and Dump Capability
- Time Management to within Instrument Specified Tolerances
- Interrupt Handling and Device Interfaces
- Memory Scrubbing (EDAC)
- Watchdog Timer Functions



Flight Software External Interfaces

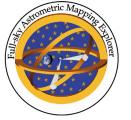


- Primarily extra-FSC interfaces as documented in:
 - ICDFM004 Spacecraft Bus to Instrument Software Interface Control Document

Uplink Bus Commands
Star Tracker Status/Data
IMU Status/Data
Sun Sensor Status/Data
Magnetometer Status/Data
Torque Rod Status
Discrete Sensors Data
Spacecraft Bus Telemetry
Uplink Instrument Commands

FAME FSW

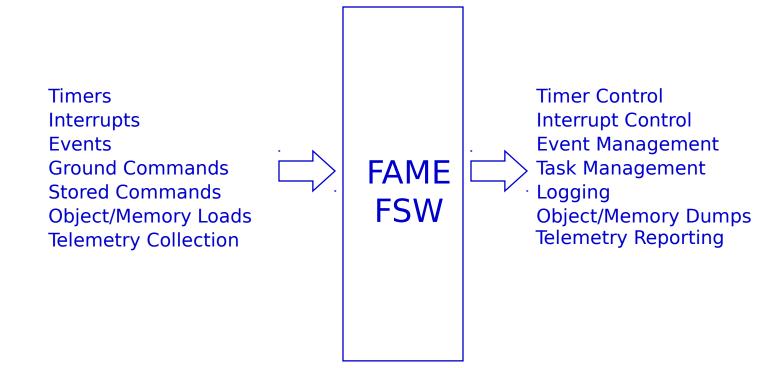
Downlink Bus Telemetry
Star Tracker Control
IMU Control
Sun Sensor Control
Instrument Commands
Spacecraft Bus Commands
Uplink Instrument Commands

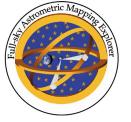


Flight Software Internal Interfaces



- Primarily intra-FSC interfaces as documented in:
 - ICDFM005 Spacecraft Bus to Flight Software Interface Control Document





Flight Software Top-Level Architecture

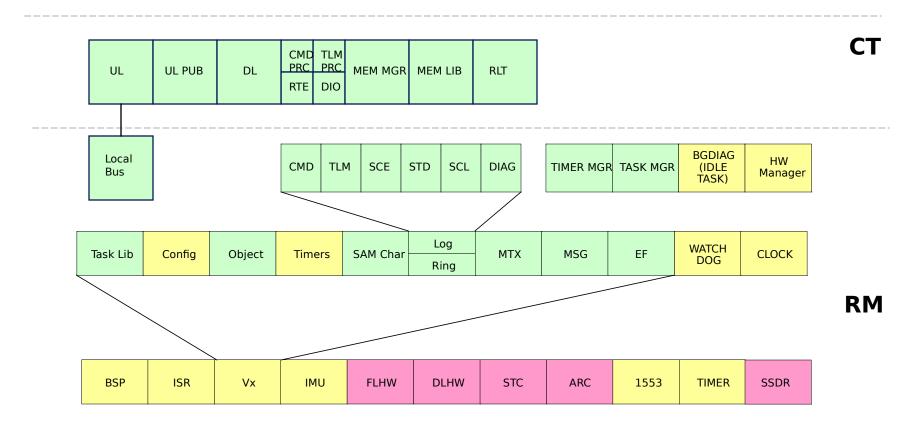


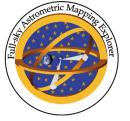
Reuse Potential



SC Stellar Compass	NAV	ADAC	ADAC EXEC	OPP	OP
STC ISR LIB		MATH LIB		ADAC CMD LIB	
ADAC SYNC		ADAC STS LIB		Orbit LIB	

ADAC





Software Development Activities



- Software Development Environment Configuration
 - VxWorks installation
 - VxSim installation
 - ClearCase CM System
 - Restore ICM FSW test hardware where applicable
- ICM Reuse Code Porting
- Uplink/Downlink design, specification, and test
- Device Drivers for currently designed FSC modules
 - Uplink/Downlink Module
 - Solid State Data Recorder Module
- Support CT&DH design
- TIM Participation
- Schedule and Budget Reviews



Reduced or Idled Software Development



- Bus/Instrument Communication Protocol development (reduced)
- Documentation Deliverables (reduced)
- GNC Requirement Specification (idle)
- GNC Design and ICM Porting (idle)
- Instrument Command and Telemetry Definitions (idle)
- Instrument Science Data Formats (idle)



IV&V Status



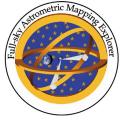
- NASA/IV&V in Fairmont will provide IV&V program management support
 - John Marinaro is the POC
- SAIC/Crystal City has been identified by NASA/IV&V as the organization that will provide IV&V engineering services
 - Paul Kirsch is the POC
- MOA Status:
 - Short term (until October 31st) MOA has been agreed to by NRL, GSFC & NASA/IV&V
 - NRL and GSFC signatures have been sent to NASA/IV&V, awaiting IV&V signatures and start of effort.
- Current agreed approach
 - Start with a short term IV&V evaluation effort (~3 months)
 - Work FAME IV&V project plan and critical function list as priorities during the initial period.
 - Preliminary review of Instrument & Bus FSW efforts will result in some initial feedback
 - Enter into long term effort that would continue until the flight software is declared ready to support the mission (revised MOA).
- Overall Scope
 - IV&V will only provide coverage for Instrument and Bus Flight Software Efforts



Software Testing



- Porting ICM Software Unit Tests as applicable
- Current Module, Integration, and System testing is not under active investigation or development



Software Documentation



NCST-SDP-FM001

Flight Software Development Plan - Currently Idle, will revise for PDR based on budget and schedule modifications

NCST-SRS-FM001

Flight Software Requirements Specification - Currently Idle, awaiting GNC inputs

NCST-ICD-FM004

Spacecraft Bus to Instrument ICD - Actively updated, including 1553 interface and bus schedule

NCST-ICD-FM005

Spacecraft Controller HW to SW ICD - Actively updated, captured baseline but awaiting any CT&DH modifications

NCST-ICD-FM003

Space to Ground ICD - Currently maintained by GSW, continuing updates of VCDU formats for supporting uplink/downlink testing

FSW CM Plan

Updated as required, captured initial CM implementation

FSW Standard Operating Procedures

Updated as required, captured initial FSW development procudures

All other documents are idle



Flight Software Builds



Build 0

System Resource Manager

Real-OS Flash Boot OS APIs Task Manager

Command and Telemetry
CCSDS C&T Formatting

Build 2

System Resource Manager

External Device Interfaces
Timer Management
Initial Boot PROMs

Command and Telemetry

Stored Commanding Limit Checking Instrument Data Processing

GNC

IMU Processing
Attitude Determination

Build 1

System Resource Manager

Breadboard H/W (FSC) Interfaces Object Loader System Logger

Command and Telemetry
Real-Time C&T Processing

Build 3

System Resource Manager

Final Boot PROMs

Command and Telemetry

Event Driven Commanding Instrument C&T Processing

GNC

Star Tracker Processing Attitude Control



Procurements



- Software Development Environment Hardware and Software has already been procured
- Awaiting Budget Decisions regarding COMET selection
- Awaiting Funding regarding VxWorks Compiler Source Code Modifications
- Continuing Software Maintenance fees provided as needed

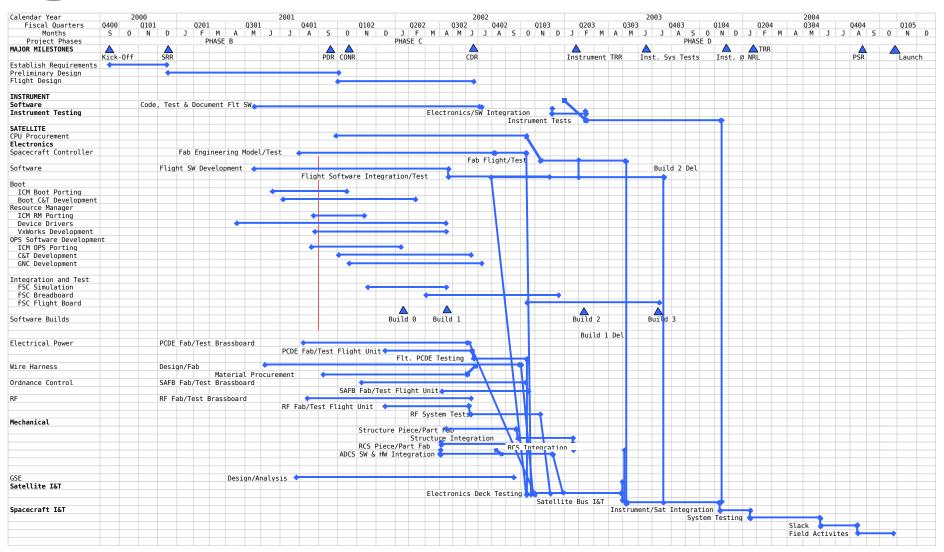


Schedule Summary











Current Open Issues



- Schedule was based on hardware design and delivery schedule that has been modified
- Testing was to be performed on hardware that may never be delivered due to budget restrictions



Backup





